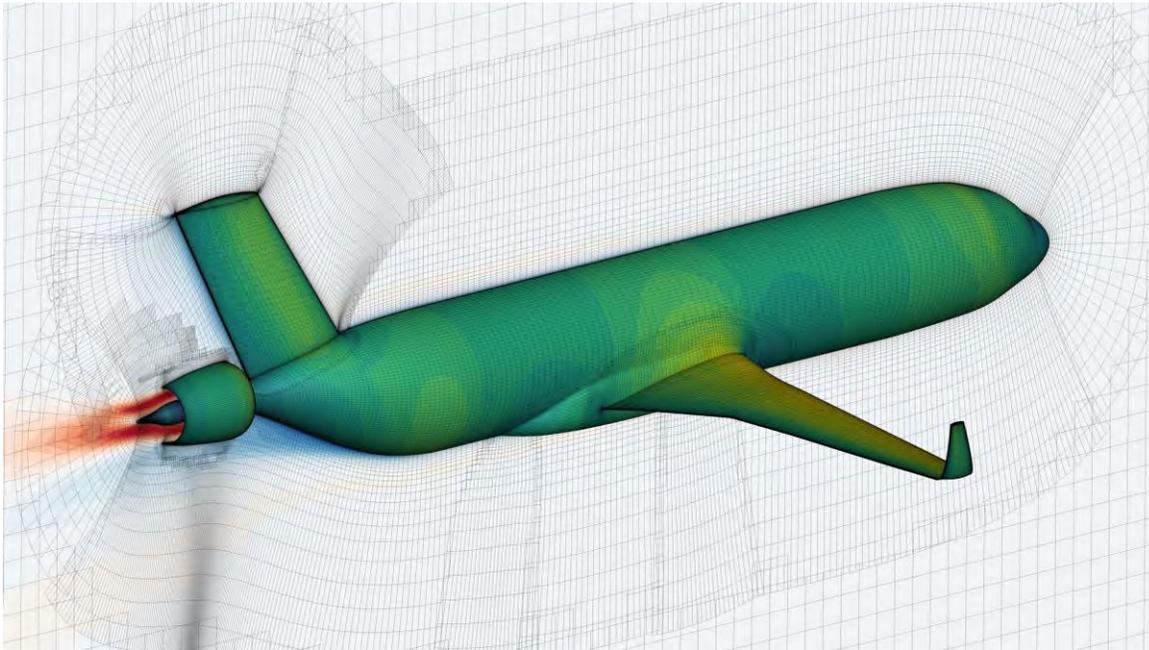


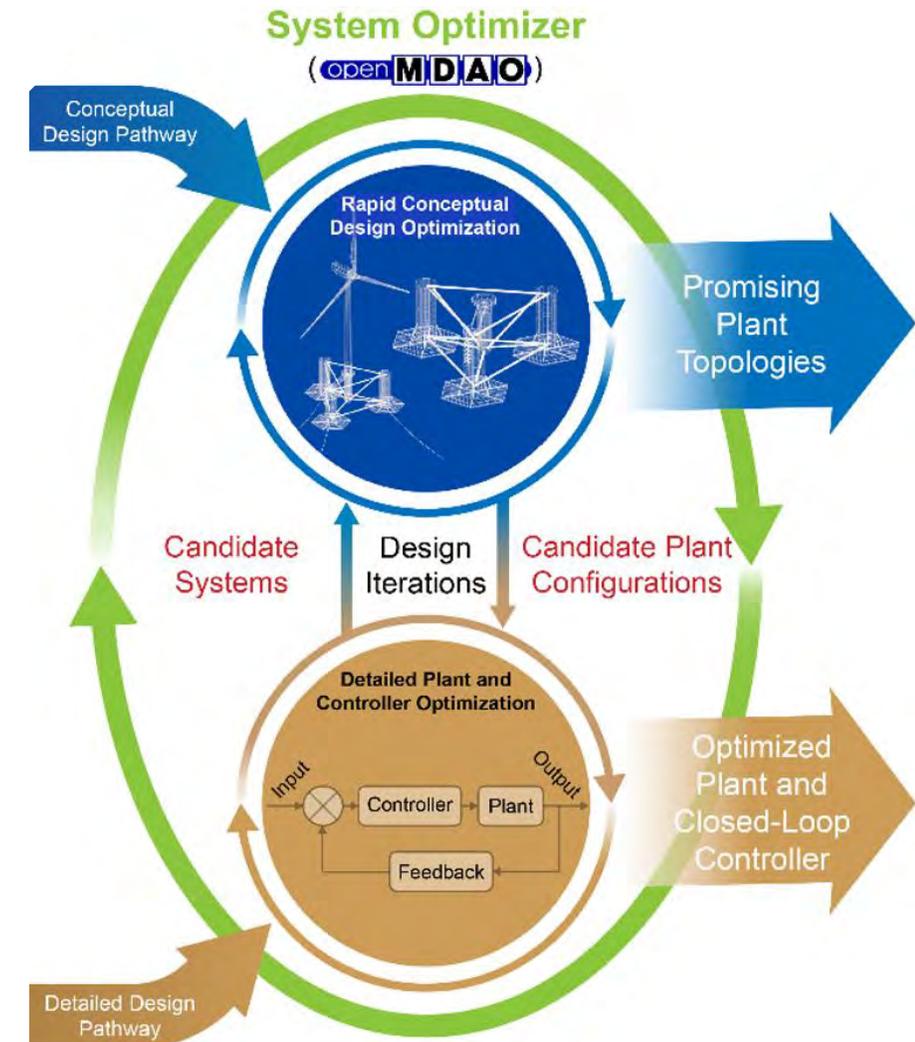
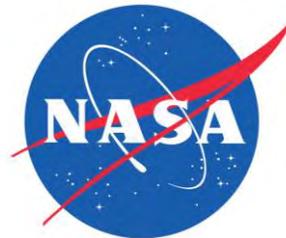
Performing multidisciplinary optimization using openMDAO



John Jasa

john.jasa@nasa.gov

NASA Glenn Research Center



What is OpenMDAO?

Who's using OpenMDAO?

Why should I use OpenMDAO?

How can I learn more about OpenMDAO?

What is OpenMDAO?

Who's using OpenMDAO?

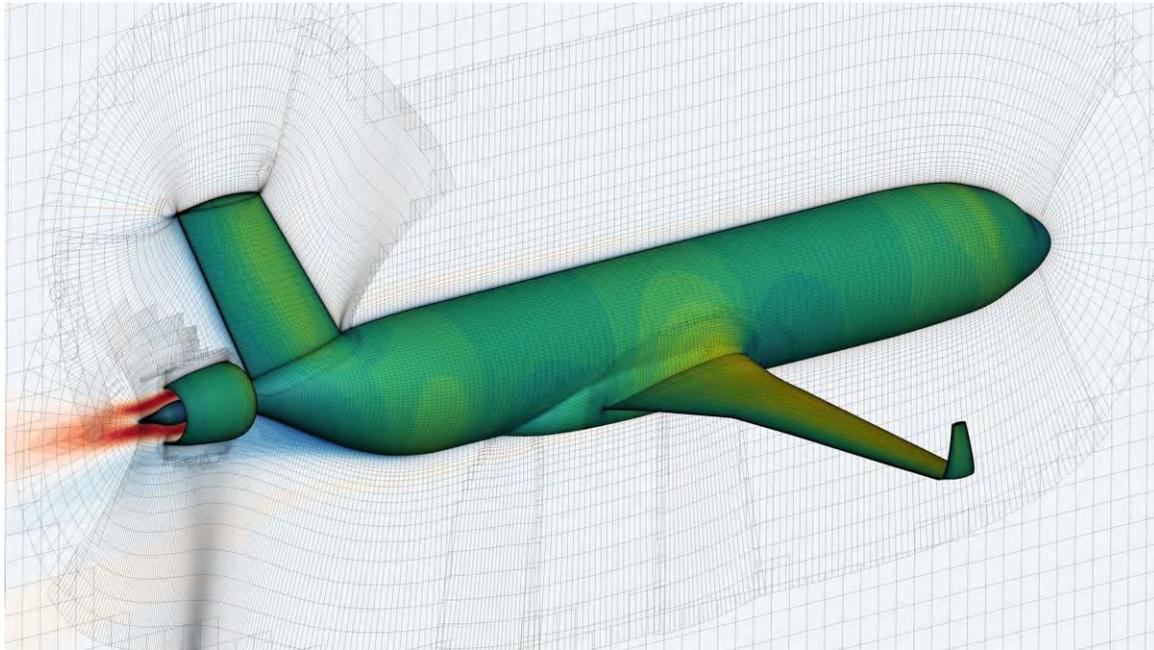
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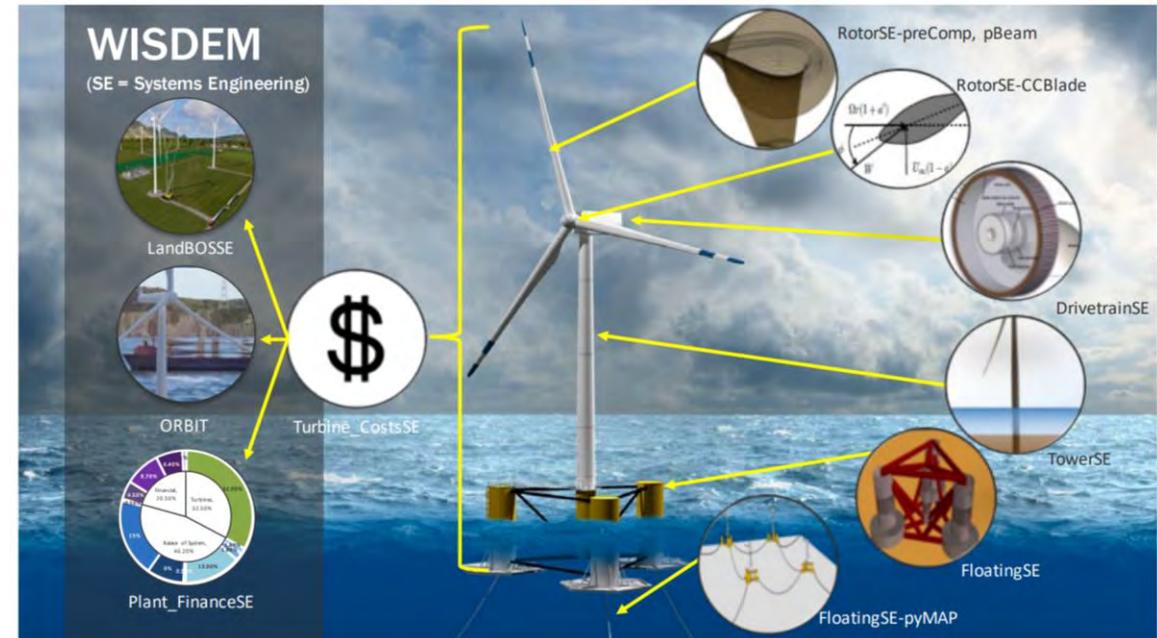
OpenMDAO is a powerful tool for doing gradient-based multidisciplinary optimization

Multidisciplinary Design Analysis and Optimization

OpenMDAO is a powerful tool for doing gradient-based multidisciplinary optimization



NASA designs complex multidisciplinary systems using OpenMDAO



WISDEM and WEIS are two NREL tools that use the OpenMDAO framework

You should use OpenMDAO if you are doing at least two of these

1

2

Multidisciplinary gradient-based optimization
using efficient derivatives in Python

3

4

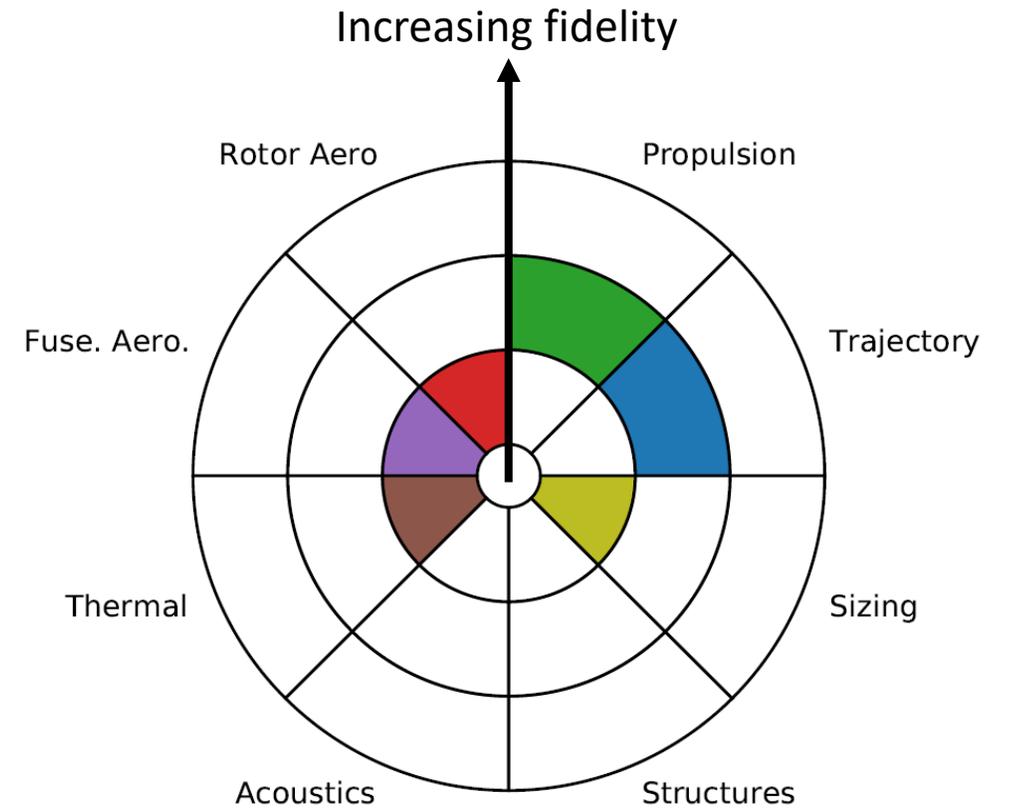
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At NASA, we use OpenMDAO to design tightly coupled aircraft



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Multidisciplinary Model					Model
Propeller	Electrical	Turboshaft	Wing	Trajectory	Analysis
OpenBEMT	ZapPy	pyCycle	OpenAeroStruct	Dymos	Library
OpenMDAO					Framework

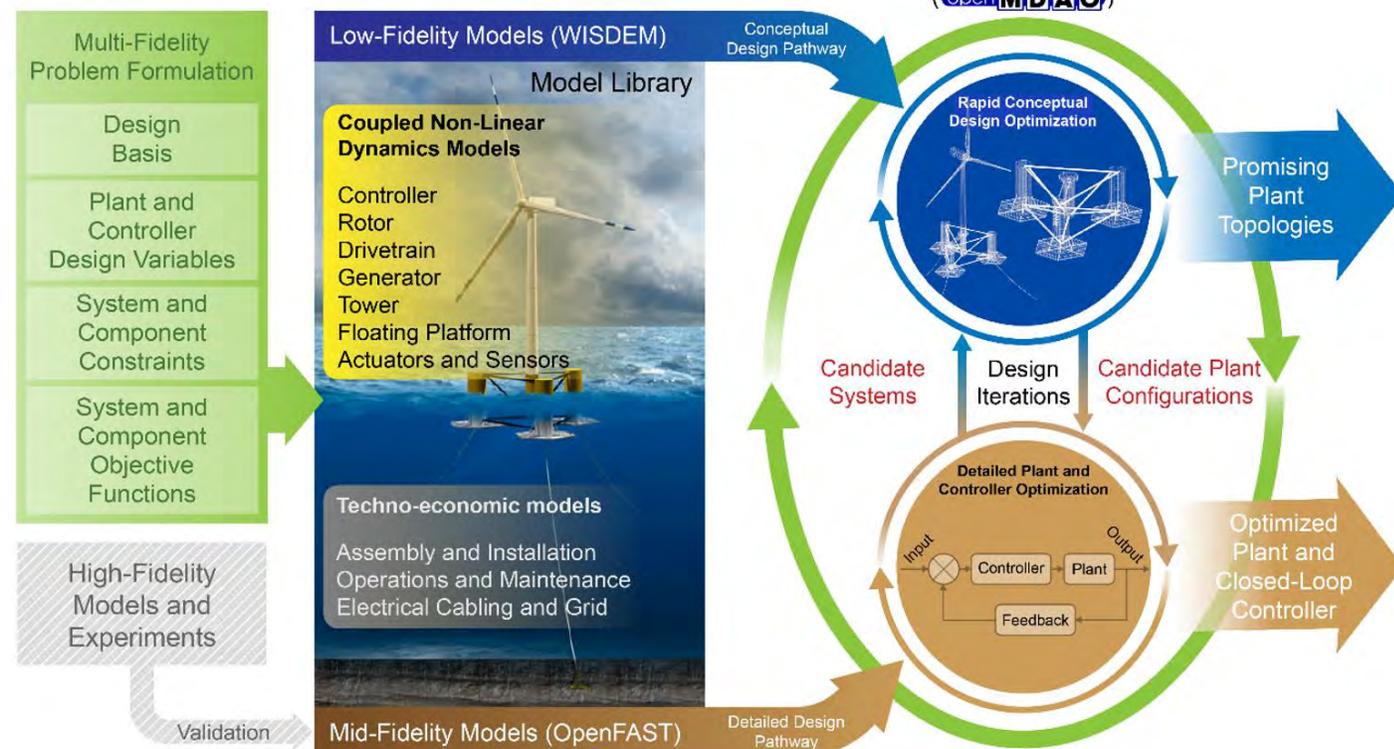
Other groups have used OpenMDAO for a variety of applications

- NREL, DOE, ARPA-E: WISDEM and WEIS wind turbine design tools
- Air Force Research Lab + Northrup Grumman: Aeropropulsive design optimization
- Aurora Flight Sciences: Aerostructural aircraft wing design
- ONREA: Aircraft and spacecraft design
- Siemens Gamesa
- Uber Elevate: Electric aircraft powertrain design
- Georgia Tech Research Institute, DOD: Model based systems engineering
- Academics: DTU, BYU, UC San Diego, University of Michigan, Georgia Tech, Stanford, MIT, RPI, Purdue, NTNU

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Wind Energy with Integrated Servo-Control (WEIS)

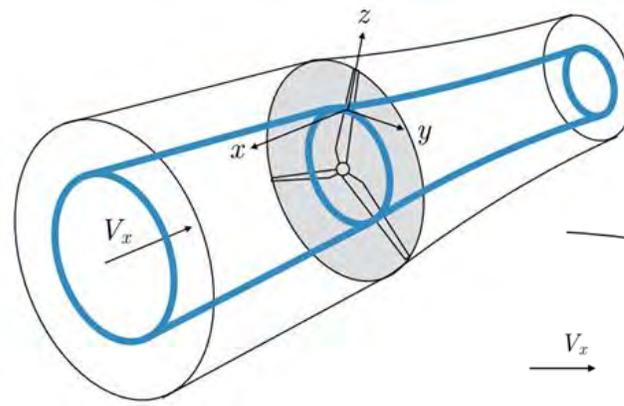


Other groups have used OpenMDAO for a variety of applications

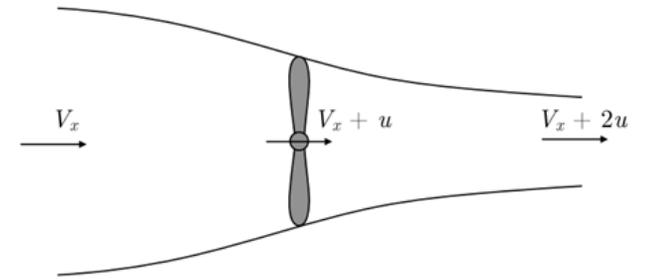
- NREL, DOE, ARPA-E: WISDEM and WEIS wind turbine design tools
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- Aurora Flight Sciences: Aerostructural aircraft wing design
- ONREA: Aircraft and spacecraft design
- Raytheon: Missile design
- Uber Elevate: Electric aircraft powertrain design
- Georgia Tech Research Institute, DOD: Model based systems engineering
- Even more! DTU, BYU, UC San Diego, University of Michigan, Georgia Tech, Stanford, MIT, RPI, Purdue, NTNU



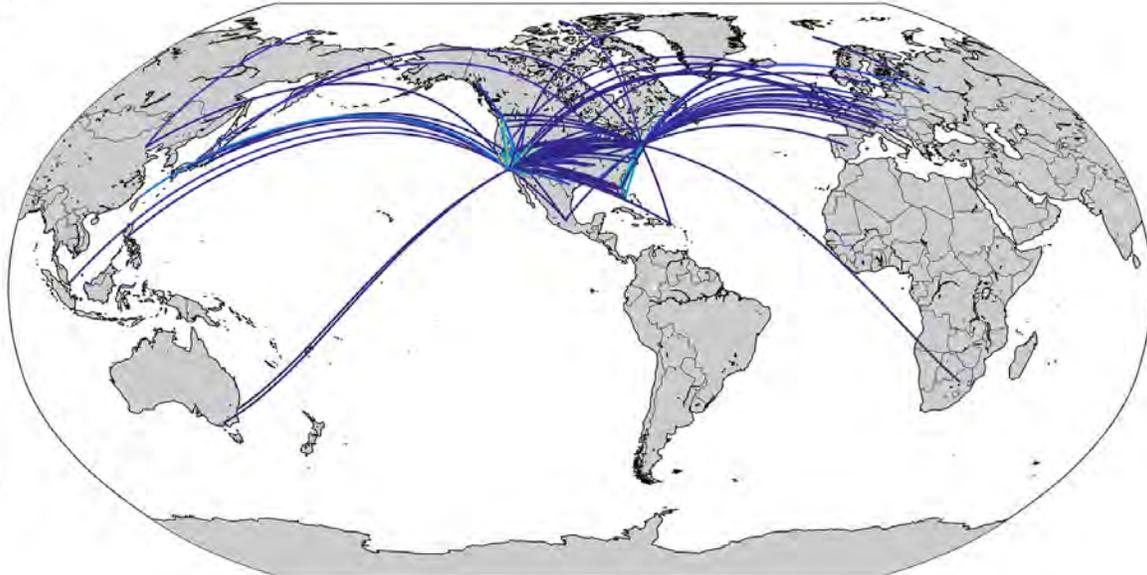
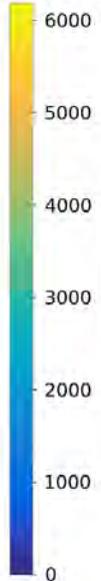
DTU



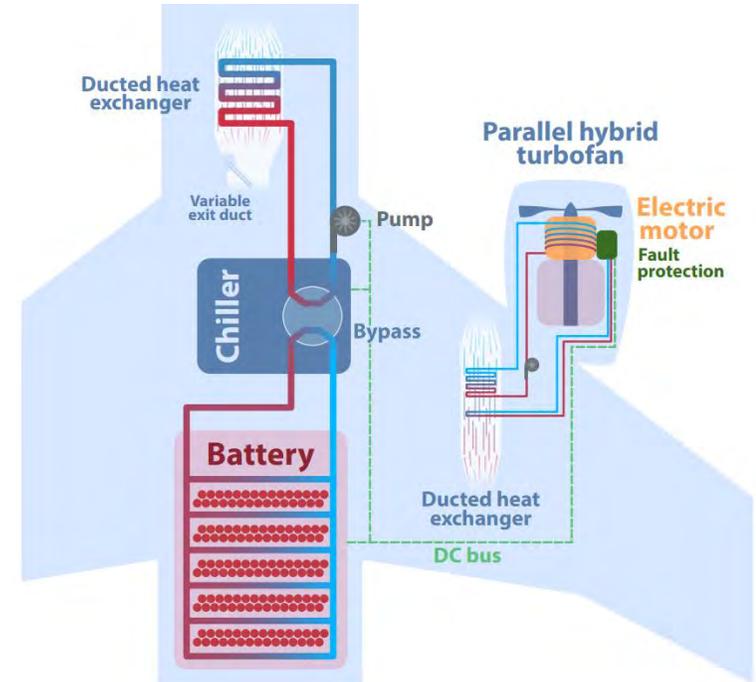
BYU, Ning 2021



Passengers per day



UC San Diego, Hwang et al 2019



U of Michigan, Adler et al 2022

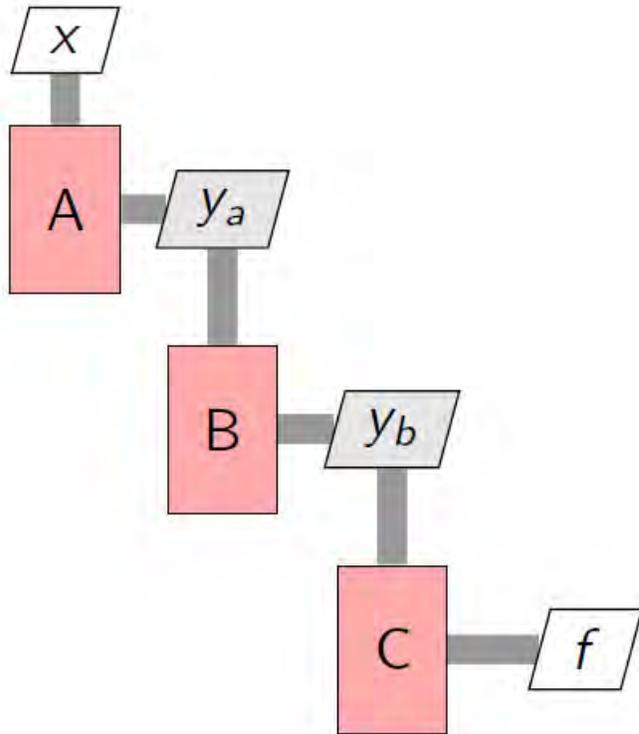
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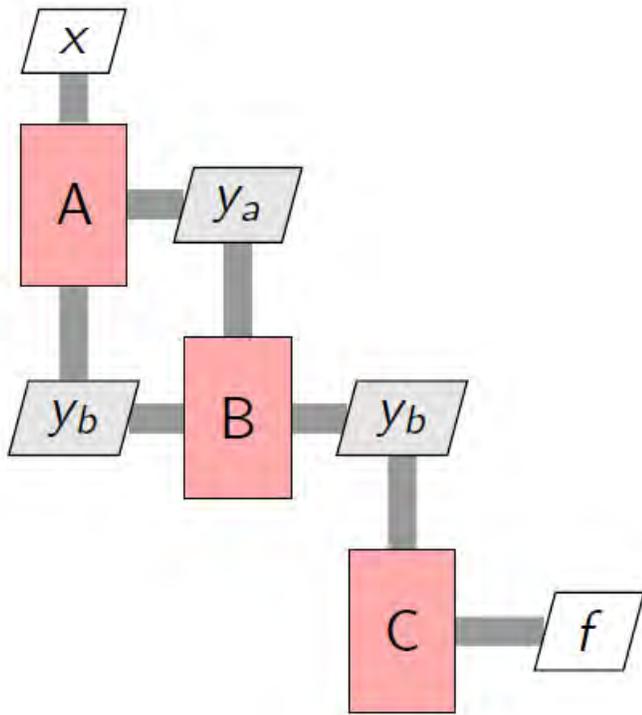
You *could* propagate derivatives through your model by hand



Derivative of f with respect to x :

$$\frac{df}{dx} = \frac{\partial f}{\partial y_b} \frac{\partial y_b}{\partial y_a} \frac{\partial y_a}{\partial x}$$

You *could* propagate derivatives through your model by hand

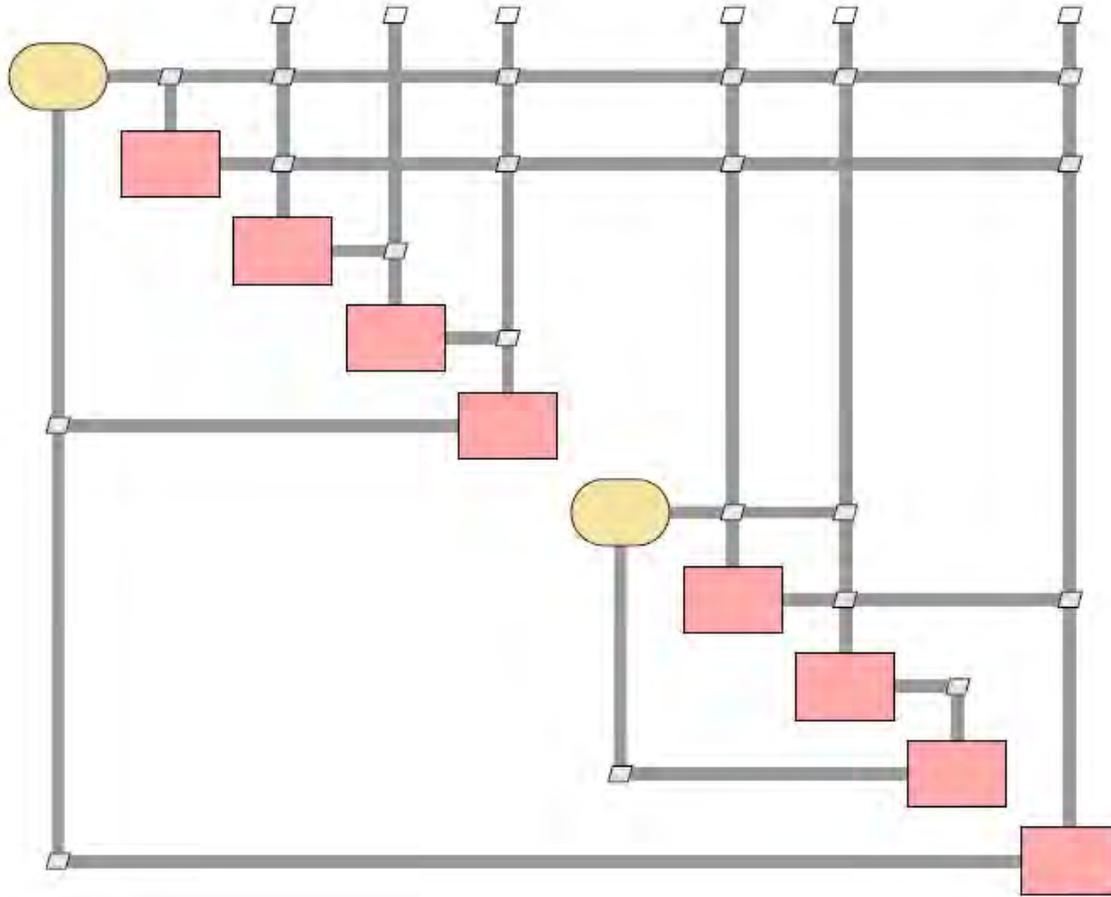


Derivative of f with respect to x :

$$\frac{df}{dx} = \frac{\partial f}{\partial y_b} \frac{\partial y_b}{\partial y_a} \frac{\partial y_a}{\partial x} + \frac{\partial f}{\partial y_b} \frac{dy_b}{dx} + \frac{\partial f}{\partial y_a} \frac{dy_a}{dx}$$

The $\frac{dy_b}{dx}$ and $\frac{dy_a}{dx}$ terms are
can be found with an adjoint solve

You *could* propagate derivatives through your model by hand but it gets untenable

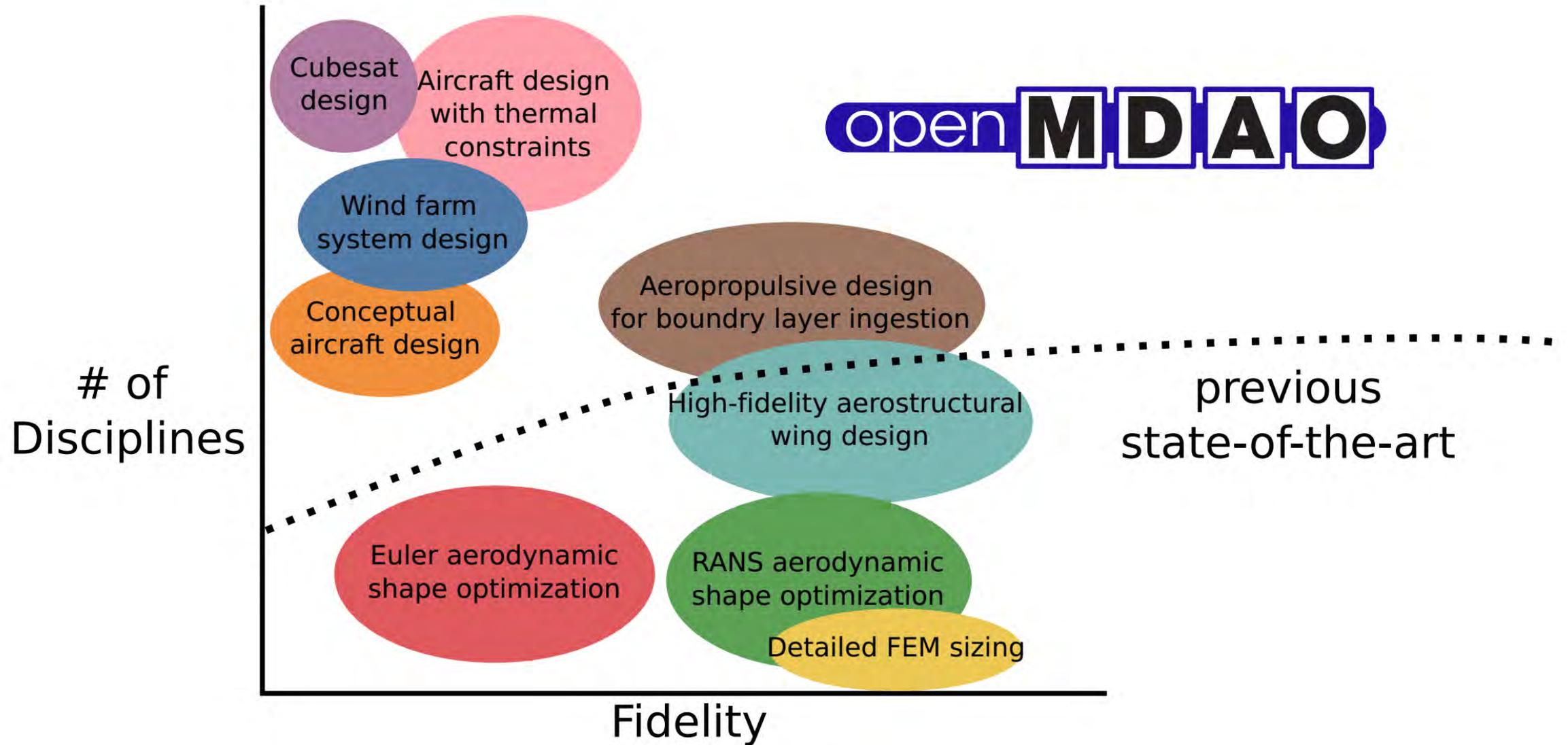


Find $\frac{df}{dx}$:

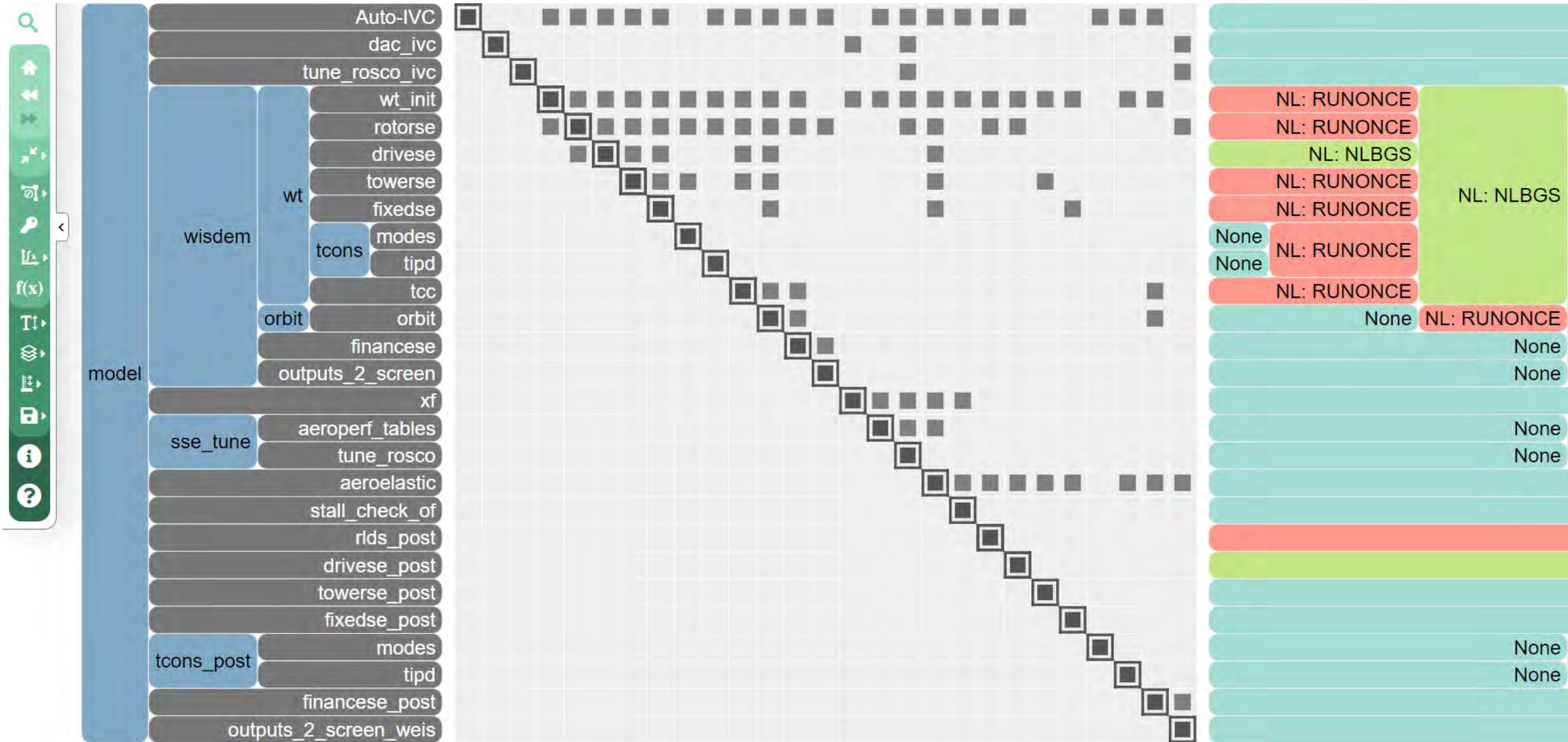
solvers within solvers??

let's be honest,
we're not doing this by hand!

Making the hard easy and the impossible hard



OpenMDAO has built-in tools to help design complicated systems



OpenMDAO has built-in tools to help design complicated systems

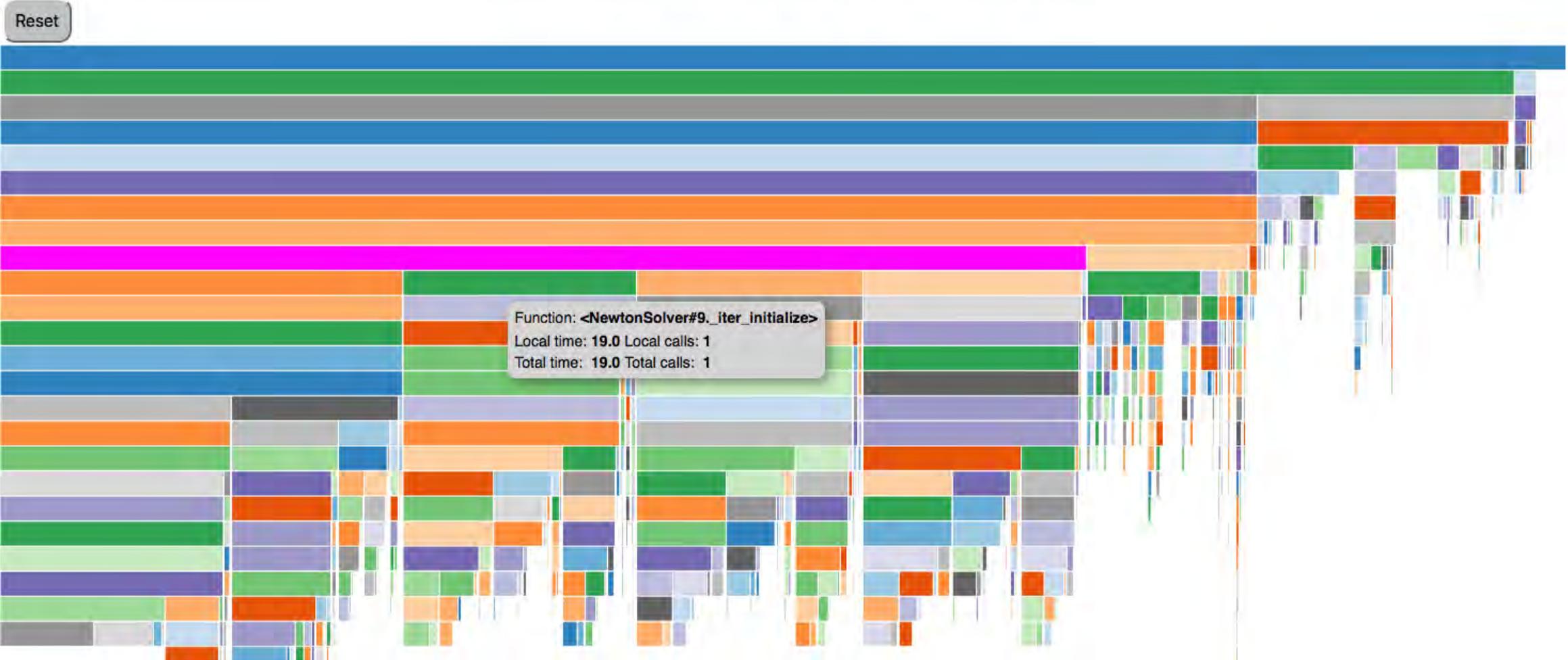
Connections for propulsor.py

Output (promoted)	Units	Value	Units	Input (promoted)
<i>design.fan.ideal_flow.n</i>		array (10,)		<i>design.fan.ideal_flow.n</i>
<i>design.fan.ideal_flow.n_moles</i>		[0.034]		<i>design.fan.ideal_flow.n_moles</i>
<i>design.fan.ideal_flow.n_moles</i>		[0.034]		<i>design.fan.ideal_flow.n_moles</i>
<i>design.fan.ideal_flow.n_moles</i>		[0.034]		<i>design.fan.ideal_flow.n_moles</i>
design.fan.press_rise.Pt_out	lbf/inch**2	[0.06895]	bar	<i>design.fan.ideal_flow.P</i>
design.fan.press_rise.Pt_out	lbf/inch**2	[0.06895]	bar	<i>design.fan.ideal_flow.P</i>
design.fan.press_rise.Pt_out	lbf/inch**2	[1.]	lbf/inch**2	<i>design.fan.ideal_flow.P</i>
design.fan.ideal_flow.props.TP2ls.lhs_TP		array (5, 5)		design.fan.ideal_flow.props.ls2p.A
design.fan.ideal_flow.props.TP2ls.rhs_P		[0. 0. 0. 0. 0.]		design.fan.ideal_flow.props.ls2p.b
design.fan.ideal_flow.props.TP2ls.lhs_TP		array (5, 5)		design.fan.ideal_flow.props.ls2t.A
design.fan.ideal_flow.props.TP2ls.rhs_T		[0. 0. 0. 0. 0.]		design.fan.ideal_flow.props.ls2t.b
design.fan.ideal_flow.props.ls2p.x		[0.1 0.1 0.1 0.1 0.1]		design.fan.ideal_flow.props.tp2props.result_P
design.fan.ideal_flow.props.ls2t.x		[0.1 0.1 0.1 0.1 0.1]		design.fan.ideal_flow.props.tp2props.result_T
<i>design.fan.ideal_flow.R</i>	(N*m)/(kg*degK)	[0.00024]	Btu/(lbm*degR)	<i>design.fan.ideal_flow.R</i>
<i>design.fan.ideal_flow.rho</i>	g/cm**3	[0.02497]	lbm/ft**3	<i>design.fan.ideal_flow.rho</i>
<i>design.fan.ideal_flow.T</i>	degK	[100.1]	degK	<i>design.fan.ideal_flow.T</i>

Absolute Outputs
 Promoted Outputs
 Output Units
 Values
 Input Units
 Absolute Inputs
 Promoted Inputs

OpenMDAO has built-in tools to help design complicated systems

Instance Profile for propulsor.py

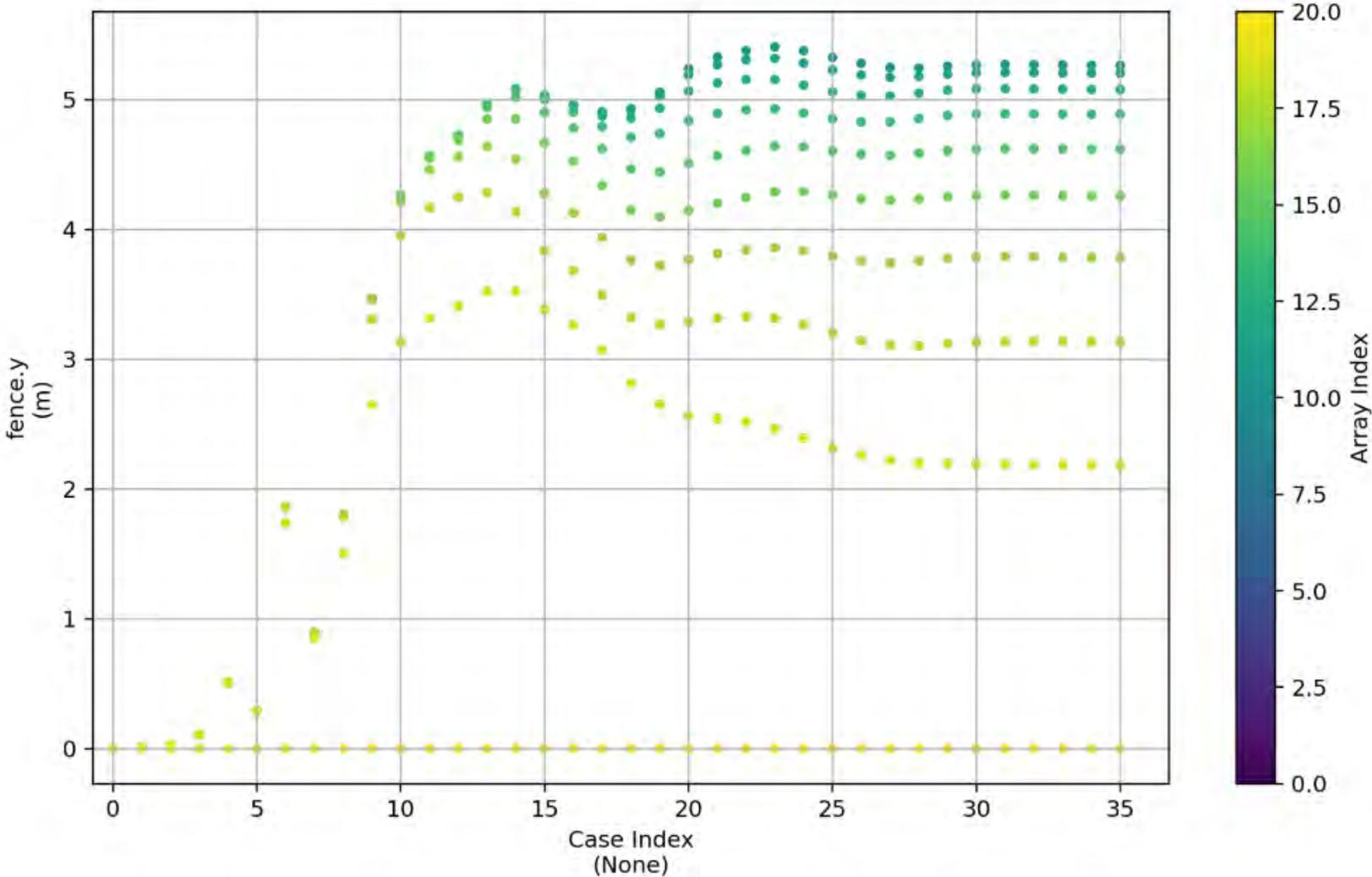


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Case #

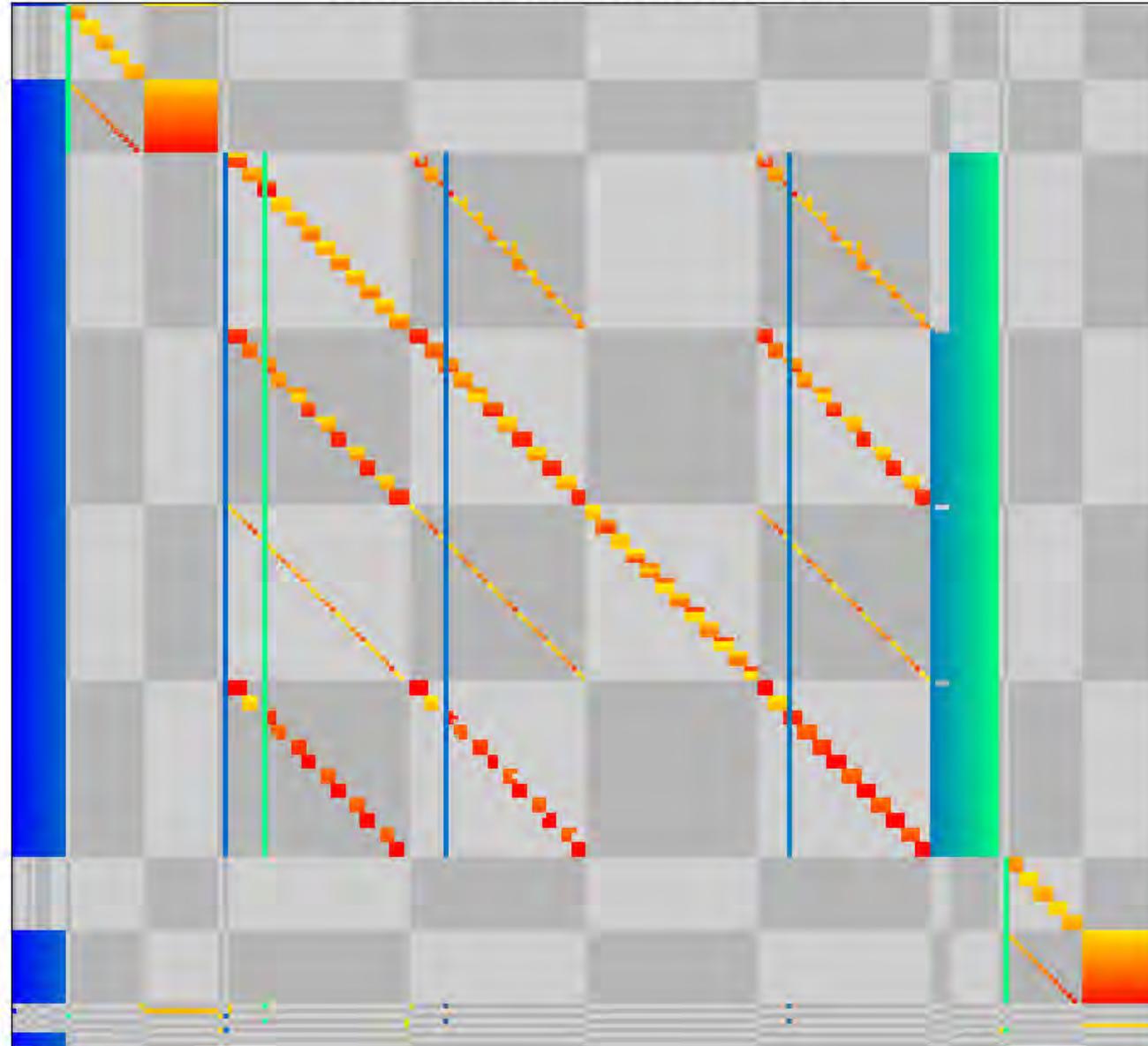
36

Figure 1



OpenMDAO has built-in tools to help design complicated systems

Total Jacobian Coloring (214 x 234)
29 fwd colors, 19 rev colors (77.6% improvement)



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More details and resources

[OpenMDAO: an open-source framework for multidisciplinary design, analysis, and optimization. Gray, Hwang et al, SMO 2019](#)

<https://openmdao.org/>

Subscribe to the [YouTube: OpenMDAO](#) and here's a link to the introductory video: <https://youtu.be/P6bFtwf485Q>

Join the mailing list: send an email with 'subscribe' in the subject to openmdao-announce-join@lists.nasa.gov

Upcoming OpenMDAO workshop

October 24-25, 2022 in Cleveland, Ohio, USA

An opportunity for users and developers to meet and discuss what they'd like to see from OpenMDAO

Sign up at openmdao.org

Thanks!

john.jasa@nasa.gov